

PNS.1 Steady Aerodynamic Flows

Farfield, subsonic-transonic potential flow, $\mathbf{u} = \nabla \cdot \phi$

DM:

$$\mathcal{L}(\phi) = (1 - M_\infty^2) \frac{\partial^2 \phi}{\partial x^2} + \frac{\partial^2 \phi}{\partial y^2} + \frac{\partial^2 \phi}{\partial z^2} = 0$$

$$\ell(\phi) = \hat{\mathbf{n}} \cdot \nabla \phi - U_\infty \hat{\mathbf{i}} \cdot \hat{\mathbf{n}} = 0$$

DE:

$$p(\mathbf{x}_\delta) = p_\infty - \rho \nabla \phi \cdot \nabla \phi / 2$$

Nearfield, boundary layers wash aerosurfaces

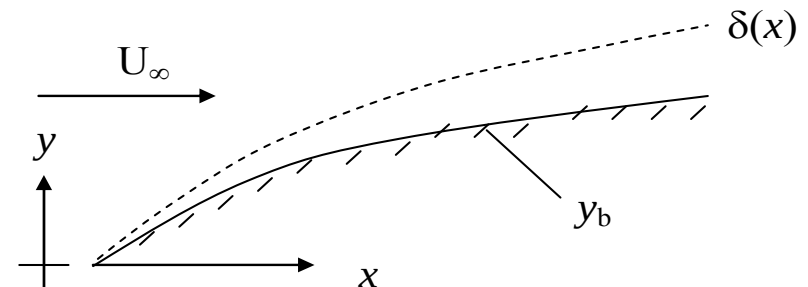
viscous, turbulent effects dominate

DM:

$$\nabla \cdot \mathbf{u} = 0$$

DP:

$$\mathbf{u} \cdot \nabla \mathbf{u} = \frac{-1}{\rho_0} \nabla p + \nabla \mathbf{T}$$



in the region $y_b(x) \leq y(x) \leq \delta(x)$

$\delta(x) \equiv$ boundary layer thickness

$\mathbf{T} =$ stress tensor (viscous + turbulent effects)

PNS.2 Parabolic Navier-Stokes, Boundary Layer Form

Reynolds ordering of steady INS $\Rightarrow n = 2$, subsonic BL

DP_y: pressure through BL is constant
 $\Rightarrow P(x)$ from potential farfield DM

DP_x: $\partial^2 u / \partial x^2$ is $O(\delta^2)$, hence negligible, $Re = O(\delta^{-2}) \gg 1$
 $\Rightarrow$ parabolic PDE on $x \geq x_0$, $0 \leq y \leq \delta(x)$

DM: $\partial v / \partial y = -\partial u / \partial x$, hence initial value on $0 < y \leq \delta(x)$

Laminar - thermal subsonic BL non-D conservation form

$$\mathcal{L}(u) = u \frac{\partial u}{\partial x} + v \frac{\partial u}{\partial y} + Eu \frac{dP_I}{dx} - \frac{1}{Re} \frac{\partial^2 u}{\partial y^2} + \frac{Gr}{Re^2} \Theta \hat{\mathbf{g}} \cdot \hat{\mathbf{i}} = 0$$

$$\mathcal{L}(\Theta) = u \frac{\partial \Theta}{\partial x} + v \frac{\partial \Theta}{\partial y} - \frac{Ec}{Re} \left(\frac{\partial u}{\partial y} \right)^2 - \frac{1}{Pe} \frac{\partial^2 \Theta}{\partial y^2} = 0$$

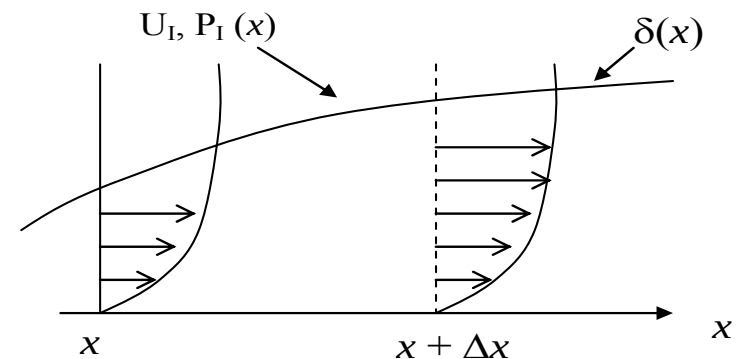
$$\mathcal{L}(v) = \frac{\partial v}{\partial y} + \frac{\partial u}{\partial x} = 0$$

BCs:

$$u(x, y = 0) = 0 = v(x, y = 0)$$

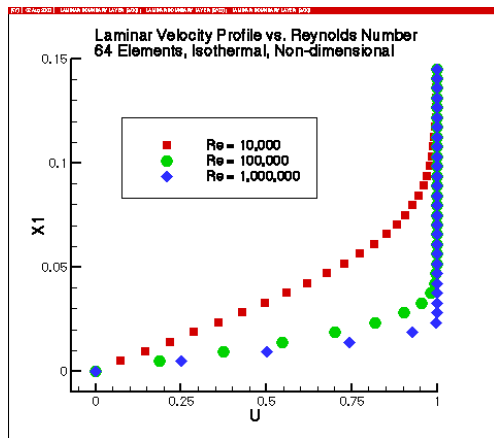
$$\left. \frac{\partial u}{\partial y} \right|_{x, y/\delta > 1} = 0, \quad \Theta(x, 0) = \Theta_{\text{wall}}$$

$$\left. \frac{\partial \Theta}{\partial y} \right|_{x, y/\delta > 1} = 0$$

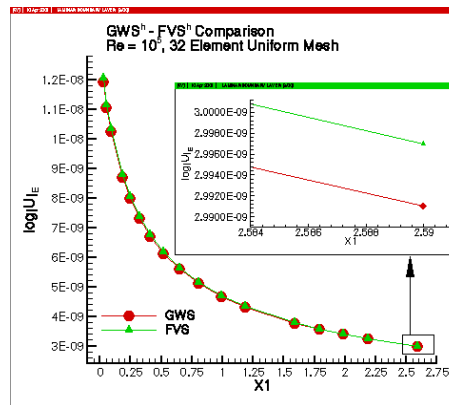


PNS.3 GWS^h , $FVS^h + \theta TS$ for BL, Solution Nuances

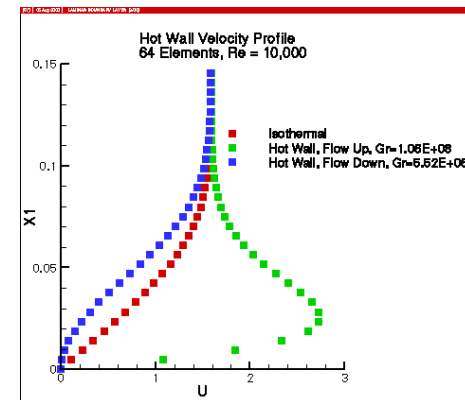
$\{U(n\Delta x)\}$ profiles for Re



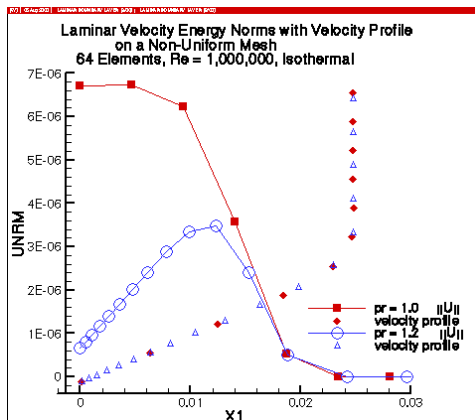
GWS^h optimality



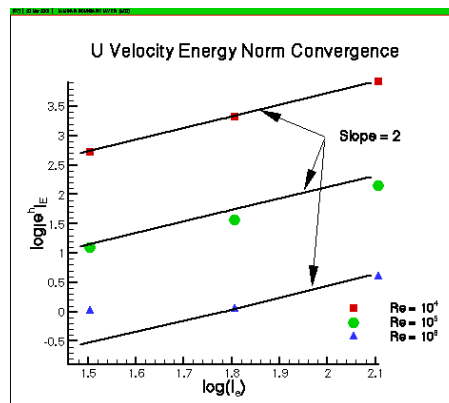
Thermal BLs



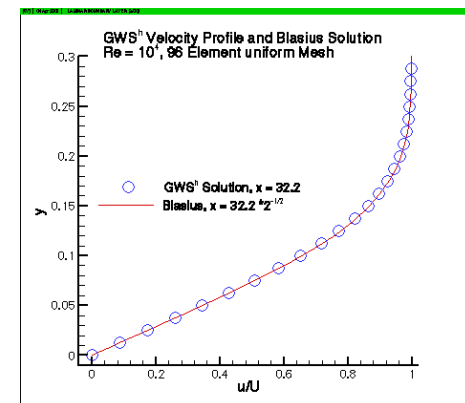
Solution mesh adaptation



GWS^h convergence



GWS^h verification



PNS.4 PNS Boundary Layer Flow, Turbulence

BL form of NS valid only for $Re \gg 1$

aircraft	Mach	U_∞ (m/s)	L (m)	Re	Re/L
commuter	0.3	125	10	3E07	$O(E06)$
wide body	0.9	250	40	2E08	$O(E06)$

BL flows will be turbulent (!)

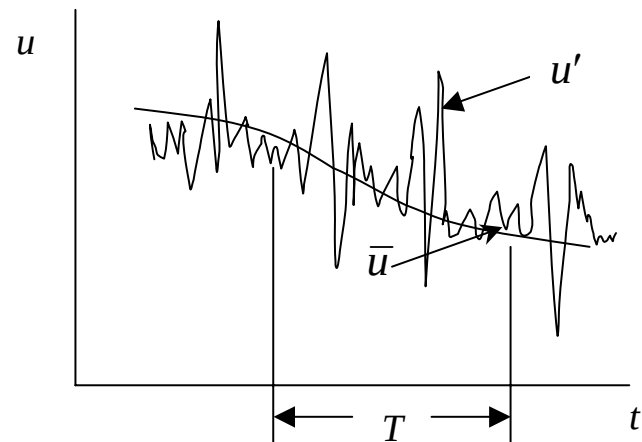
resolution of BL velocity components

$$u(\mathbf{x}, t) \equiv \bar{u}(\mathbf{x}) + u'(\mathbf{x}, t)$$

time-averaging

$$\bar{u}(\mathbf{x}) \equiv \lim_{T \rightarrow \infty} \frac{1}{T} \int_{t_0}^{t_0+T} u(\mathbf{x}, \tau) d\tau$$

$$\overline{u'} = 0$$



laboratory data

PNS.5 Turbulent Boundary Layer, Reynolds Stress

Time averaging of BL DM and DP

DM: both terms linear, hence $\nabla \cdot \bar{\mathbf{u}} = 0 = \nabla \cdot \mathbf{u}'$

DP_x: non-linear convection term generates a new contribution

$$u \frac{\partial u}{\partial x} + v \frac{\partial u}{\partial y} \Rightarrow \frac{\partial}{\partial x}(uu) + \frac{\partial}{\partial y}(vu) \quad \text{via DM}$$

$$\overline{uu} \Rightarrow \bar{u} \bar{u} + \overline{u'u'}$$

$$\overline{vu} \Rightarrow \bar{v} \bar{u} + \overline{v'u'}$$

Reynolds ordering confirms that $O(\overline{u'u'}) \approx O(\overline{v'u'}) \approx O(\delta)$

$$\frac{\partial}{\partial x} (\bar{u} \bar{u} + \overline{u'u'}) \Rightarrow O(1 \cdot 1 / 1 + \delta / 1)$$

$$\frac{\partial}{\partial y} (\bar{v} \bar{u} + \overline{v'u'}) \Rightarrow O(\delta \cdot 1 / \delta + \delta / \delta)$$

hence:

Reynolds normal stress $\overline{u'u'}$ contribution negligible

Reynolds shear stress $\overline{v'u'}$ contribution must be included

PNS.6 Boundary Layer Flow, Turbulence Modeling

Reynolds kinematic shear stress modeled after Stokes

$$\overline{v'u'} \equiv -v^t \frac{\partial \bar{u}}{\partial y}, \quad v^t \equiv \text{turbulent "eddy" viscosity, units } (\mu/\rho_\infty = \nu) \Rightarrow (L^2/t)$$

Prandtl mixing length model

$$v^t \equiv (\omega \ell_m)^2 \left| \frac{\partial \bar{u}}{\partial y} \right| f \Rightarrow (L^2)(1/t)$$

where: $\ell_m \equiv$ mixing length
 $\omega, f =$ near wall, freestream damping

Turbulent kinetic energy-dissipation model

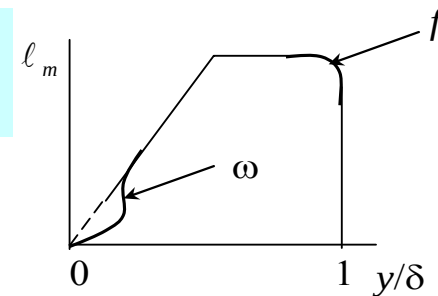
$$v^t \equiv C_\mu k^2 / \varepsilon \Rightarrow (L/t)^4 (t^3/L^2)$$

where:

$$k \equiv \frac{1}{2} \overline{(\mathbf{u}' \cdot \mathbf{u}')} = \frac{1}{2} \overline{(u'u' + v'v' + w'w')}$$

$$\varepsilon \equiv \frac{2\nu}{3} \overline{\left(\frac{\partial u'_i}{\partial x_k} \frac{\partial u'_i}{\partial x_j} \right)} \delta_{jk}$$

and: $\mathcal{L}(k)$ and $\mathcal{L}(\varepsilon)$ BL forms augment BL DM & DP_x



PNS.7 GWS^h + θTS, Turbulent BL, MLT Closure

Turbulent BL conservation law form, time-averaged $q(x,y)$, MLT

$$\text{DP}_x : \mathcal{L}(\bar{u}) = \bar{u} \frac{\partial \bar{u}}{\partial x} + \bar{v} \frac{\partial \bar{u}}{\partial y} - \frac{1}{\text{Re}} \frac{\partial}{\partial y} (1 + \text{Re}^t) \frac{\partial \bar{u}}{\partial y} + \frac{dP^I}{dx} + \frac{\text{Gr}}{\text{Re}^2} \bar{\Theta} \hat{\mathbf{g}} \cdot \hat{\mathbf{i}} = 0$$

$$\text{D}\Theta : \mathcal{L}(\bar{\Theta}) = \bar{u} \frac{\partial \bar{\Theta}}{\partial x} + \bar{v} \frac{\partial \bar{\Theta}}{\partial y} - \frac{1}{\text{Re}} \frac{\partial}{\partial y} \left(\frac{1}{\text{Pr}} + \frac{\text{Re}^t}{\text{Pr}^t} \right) \frac{\partial \bar{\Theta}}{\partial y} - \frac{\text{Ec}}{\text{Re}} \left(\frac{\partial \bar{u}}{\partial y} \right)^2 \equiv 0$$

$$\text{DM} : \mathcal{L}(\bar{v}) = \frac{\partial \bar{v}}{\partial y} + \frac{\partial \bar{u}}{\partial x} = 0$$

$$\text{DP}_y : \mathcal{L}(\bar{p}) = \frac{\partial}{\partial y} \left(\rho_0 P^I + \overline{v'v'} \right) = 0$$

where: $\text{Re}^t \equiv (v^t/v)_{\text{dim}} = \text{turbulent Reynolds number}$

$$v^t \equiv (\omega \ell_m)^2 \left| \frac{\partial \bar{u}}{\partial y} \right| f = \text{MLT eddy viscosity}$$

$\omega = 1 - \exp(-y/A) = \text{van Driest damping, } A \approx 25$

$$\ell_m = \text{Prandtl mixing length} = \begin{cases} \kappa y, & \text{on } 0 \leq y/\delta \leq \lambda/\kappa \\ \lambda \delta, & \text{on } \lambda/\kappa < y/\delta \leq 1 \end{cases} \begin{cases} \kappa = 0.405 \\ \lambda = 0.09 \end{cases}$$

$$f = [1 + 5.5(y/\delta)^6]^{-1} = \text{Klebanoff damping}$$

$\text{Pr}^t \cong \text{Pr}$ for turbulent Prandtl number (usually)

$\overline{v'v'} = \text{Reynolds transverse normal stress}$

PNS.8 GWS^h + θ TS Performance, Turbulent BL, MLT Closure

Accuracy, convergence, *regular non-uniform* Ω^h refinement

theory:
$$\left| e^h(n\Delta x) \right|_E \leq C h_e^{2\gamma} \|\text{data}\|_{H^{k-1}}^2 + C_x \Delta x^3 \|U_0\|_{H^1}^2, \gamma = \min(k, r-1)$$

norm:
$$\left| u^h(n\Delta x) \right|_E \equiv \frac{1}{2} \int_{\Omega} v^t \left(\frac{\partial u^h}{\partial y} \right)^2 dy = \frac{1}{2} \sum_{\Omega^h} \int_{\Omega_e} (\cdot) dy$$

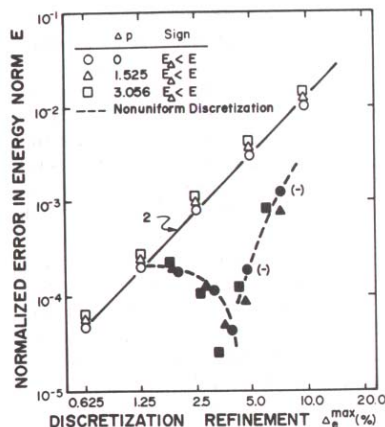
$$= \frac{1}{2 \text{Re}} \sum_e^M \{U\}_e^T \{\text{RET}\}_e^T [\text{A3011}] \{U\}_e$$

IC, M = 80 laminar

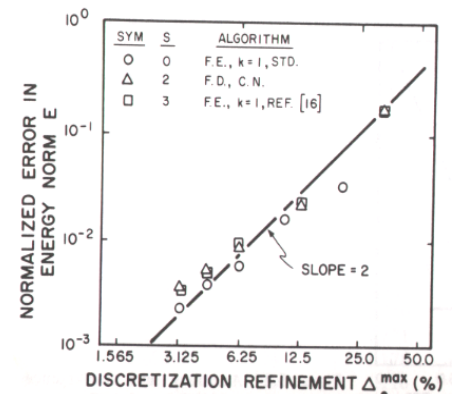
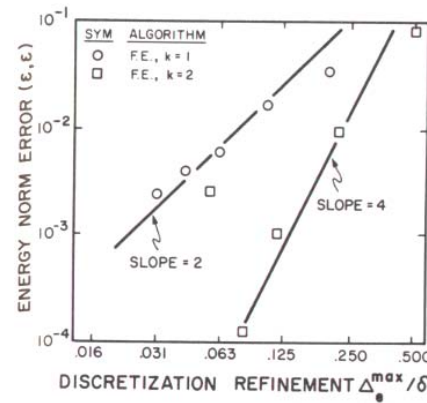
Ω^h progressions

Convergence

Optimality



Linear basis, $k=1$		Quadratic basis, $k=2$	
M	ρ	M	ρ
12	1.627	6	2.814
24	1.222	12	1.510
36	1.125	18	1.271
48	1.083	24	1.176
60	1.061	30	1.110



PNS.9 Turbulent Boundary Layer, TKE Closure

Turbulent kinetic energy-isotropic dissipation closure model

eddy viscosity:

$$\nu^t \equiv C_\mu k^2 / \varepsilon, \quad C_\mu = 0.09$$

$$k \equiv \frac{1}{2} (\overline{\mathbf{u}' \cdot \mathbf{u}'}) = \frac{1}{2} (\overline{u'u'} + \overline{v'v'} + \overline{w'w'}) \quad \varepsilon \equiv \frac{2\nu}{3} \left(\overline{\frac{\partial u'_i}{\partial x_k} \frac{\partial u'_i}{\partial x_j}} \right) \delta_{jk}$$

$\mathcal{L}(k, \varepsilon)$ conservation PDEs, non-D BL form

$$\mathcal{L}(k) = u \frac{\partial k}{\partial x} + v \frac{\partial k}{\partial y} - \frac{1}{\text{Pe}} \frac{\partial}{\partial y} \left(1 + \frac{\text{Re}^t}{C_k} \right) \frac{\partial k}{\partial y} - \tau_{12} \frac{\partial u}{\partial y} + \varepsilon = 0$$

$$\mathcal{L}(\varepsilon) = u \frac{\partial \varepsilon}{\partial x} + v \frac{\partial \varepsilon}{\partial y} - \frac{1}{\text{Pe}} \frac{\partial}{\partial y} \left(1 + \frac{\text{Re}^t}{C_\varepsilon} \right) \frac{\partial \varepsilon}{\partial y} - C_\varepsilon^1 \frac{\varepsilon}{k} \tau_{12} \frac{\partial u}{\partial y} + C_\varepsilon^2 \frac{\varepsilon}{k} \varepsilon = 0$$

$$\mathcal{L}(\tau_{12}) = \tau_{12} + \nu^t \frac{\partial u}{\partial y} = \tau_{12} + C_\mu \frac{k^2}{\varepsilon} \frac{\partial u}{\partial y} = 0$$

TKE model adds non-linear parabolic PDE + BCs + IC pair

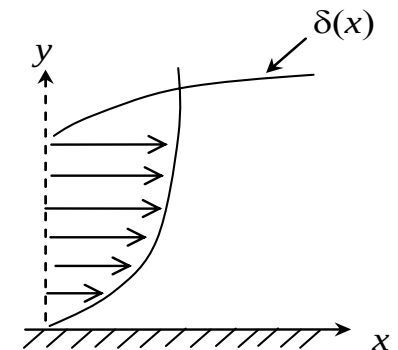
BCs:

$$k(x, y=0) = 0, \quad \varepsilon(x, y=0) \Rightarrow \varepsilon_w < \infty$$

$$\left. \frac{\partial k}{\partial y}, \frac{\partial \varepsilon}{\partial y} \right|_{y=\delta(x)} = 0$$

IC:

$$k(x_0, y) = ? = \varepsilon(x_0, y)$$



PNS.10 TKE for Turbulent BL, Near-Wall Corrections

TKE closure model requires near-wall corrections

low Re^t closure model constant modifications (Lam-Bremhorst)

$$v^t \Rightarrow f_v C_\mu k^2 / \varepsilon : f_v = \left(1 - \exp(-0.0165 R_y)\right)^2 \left(1 + 20.5 / Re^t\right)$$

$$C_\varepsilon^1 \Rightarrow f_k C_\varepsilon^1 : f_k = (1 + 0.05 f_v^{-1})^3$$

$$C_\varepsilon^2 = f_\varepsilon C_\varepsilon^2 : f_\varepsilon = 1 - \exp(-Re^t)^2$$

$$Re^t = v^t / \nu$$

$$R_y = k^{1/2} y / \nu$$

BL similarity TKE variable distributions as $f(y)$

$$U^+ \equiv U / u_\tau = \kappa^{-1} \log(y^+ E) + B$$

$$y^+ \equiv u_\tau y / \nu$$

near-wall production = dissipation in $\mathcal{L}(k)$

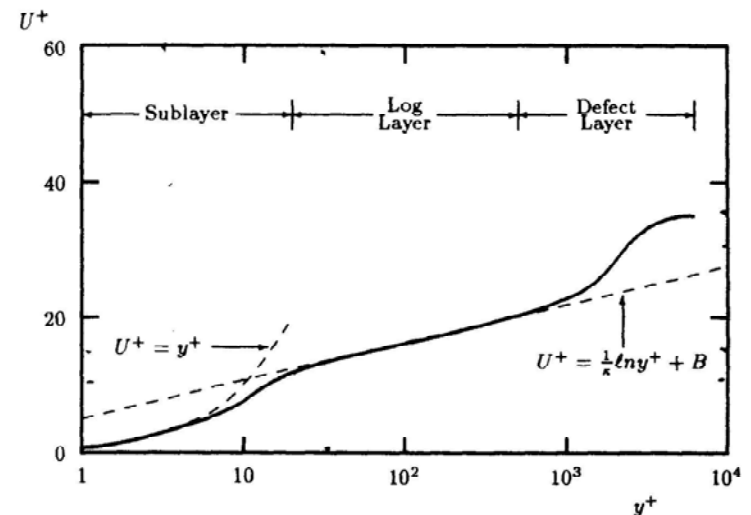
$\Rightarrow$

$$v^t = \kappa y u_\tau$$

$$k = u_\tau^2 / C_\mu$$

$$\varepsilon = (\kappa y)^{-1} |u_\tau|^3$$

$$\tau_w = \sqrt{C_\mu} k = u_\tau (C_\mu)^{-1/2}$$



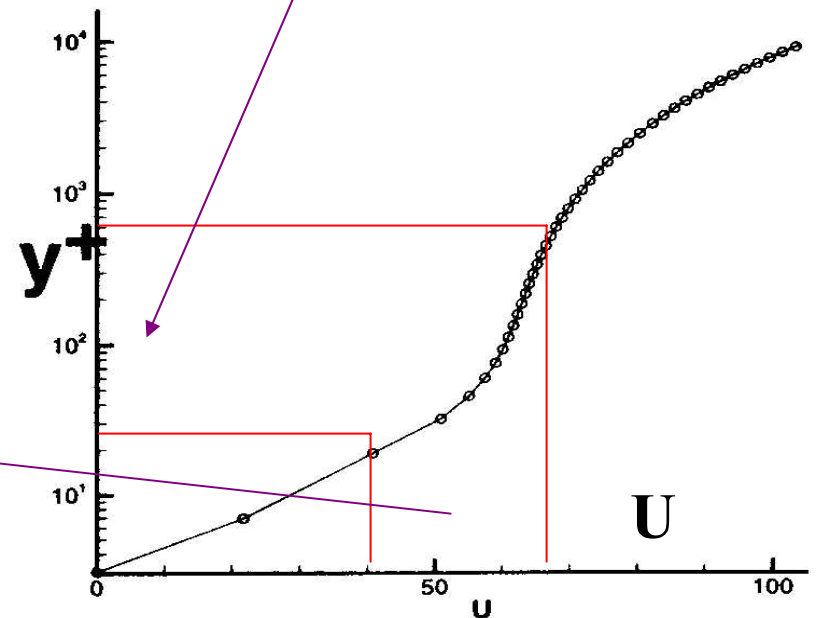
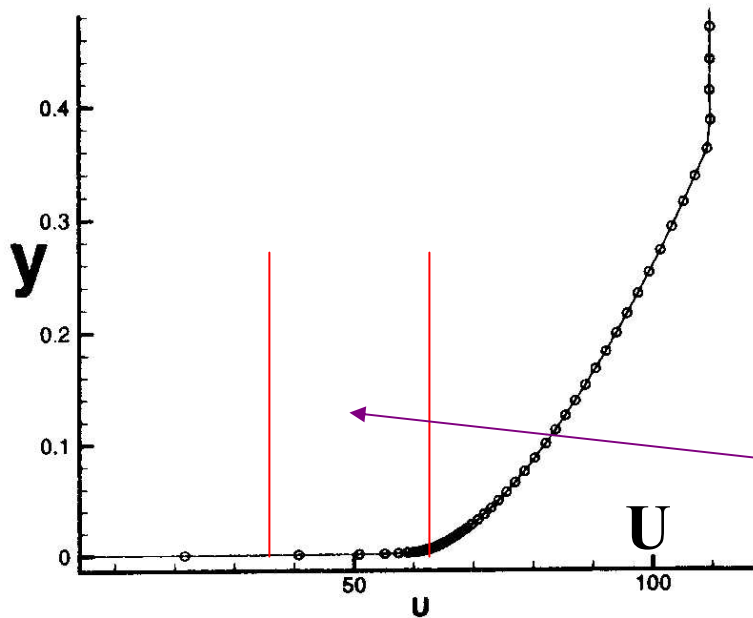
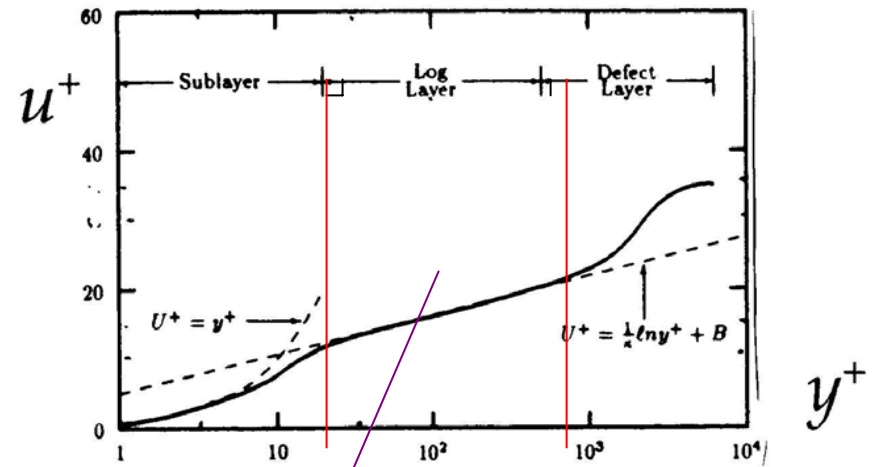
PNS.10A Turbulent Boundary Layer Similarity

Turbulent BL similarity

$$U/u_\tau = \frac{1}{\kappa} \ln(y^+ E) + B \equiv U^+$$

$$y^+ = u_\tau y / \nu$$

example: IDENT 2400 boundary layer



PNS.11 $GWS^h + \theta TS$ Template, BL, TKE + Low Re^t

Template pseudo-code modifications ($\{FV\}_e$ unchanged)

$$\text{for } \{Q\}_e = \begin{Bmatrix} U \\ T \\ K \\ E \end{Bmatrix}_e :$$

$$\begin{aligned} \{FQ\}_e = & () () \{ \overline{U} \} (1) [A3000] \{QP - QN\} \\ & + (\Delta x / 2) () \{VP, VN\} (0) [A3001] \{QP, QN\} \\ & + (\Delta x / 2, Pa^{-1}) () \{RET\} [A3011] \{QP, QN\} + \{b(Q)\} \end{aligned}$$

Source terms $\{b(\cdot)\}$ unchanged for $\{Q\} = \{U, T\}^T$, and

$$\begin{aligned} \{b(K)\}_e &= \int_{\Omega_e} \{N\} (\tau_{12} \partial u_e / \partial y) dy + \int_{\Omega_e} \{N\} \varepsilon_e dy \\ &= (\Delta x / 2) () \{TXY\} (0) [A3001] \{U\} + (\Delta x / 2) () \{ \} (1) [A200] \{E\} \\ \{b(E)\}_e &= C_\varepsilon^1 \int_{\Omega_e} \{N\} (\tau_{12} (\varepsilon / k)_e \partial u_e / \partial y) dy + C_\varepsilon^2 \int_{\Omega_e} \{N\} (\varepsilon / k)_e \varepsilon_e dy \\ &= (\Delta x / 2, CE1) (FE1) \{TXY, E / K\} (0) [A3001] \{E\} \\ &+ (\Delta x / 2, CE2) (FE2) \{E / K\} (1) [A3000] \{E\} \end{aligned}$$

Reynolds algebraic shear stress model template

$$\begin{aligned} \{FTXY\}_e &= () () \{ \} (1) [A200] \{TXY\} \\ &+ (Re^{-1}) (FNU) \{RET\} (0) [A3001] \{U\} \end{aligned}$$

PNS.12 GWS^h + θ TS TKE BL, Quasi-Newton Jacobian

Size, deeply embedded non-linearity precludes Newton

quasi-Newton
jacobians:

$$\begin{bmatrix} JUU & JUV & JUT \\ JVU & JVV & 0 \\ JTU & JTV & JTT \end{bmatrix}_e, \begin{bmatrix} JKK & JKE & JKT_{xy} \\ JEK & JEE & JET_{xy} \\ JT_{xy}K & JT_{xy}E & JT_{xy}T_{xy} \end{bmatrix}_e$$

solution sequence:

$\{\delta U, \delta V, \delta T\}^{p+1}$ unchanged from laminar, MLT
update $\{U, V, T\}^{p+1}$
 $\{\delta K, \delta E, \delta T_{xy}\}^{p+1}$ uses $\{U, V, T\}^{p+1}$
update $\{K, E, T\}^{p+1}$
index p , return to $\{\delta U, \delta V, \delta T\}^{p+1}$

oscillating convergence:

use $\{RETN\}$ in $\{FU, FT\}^p$
use $\{UN, VN\}$ in $\{FK, FE, FT_{xy}\}^p$

templates:

$[JAC]_e$ for $\{FU, FV, FT\}$ are unchanged
 $[JAC]_e$ for $\{FK, FE, FT_{xy}\}$ fully utilizes chain rule

PNS.12A GWS^h + θTS TKE Closure Jacobian Coupling

Jacobian coupling for convection terms is unchanged

$$\text{diffusion term: } \mathcal{L}(q) \Rightarrow -\frac{1}{\text{Re}} \frac{\partial}{\partial y} \left(\frac{1}{\text{Pr}} + \frac{\text{Re}^t}{C_q \text{Pr}^t} \right) \frac{\partial q}{\partial y}, \quad q = \{k, \varepsilon\}$$

$$\frac{\partial}{\partial q}(\cdot) = \frac{1}{\text{Re}} \frac{\partial}{\partial y} \left(1 + \frac{\text{Re}^t}{C_q} \right) \frac{\partial(\cdot)}{\partial y} + \frac{1}{\text{Re}} \frac{\partial}{\partial y} \left(\frac{\partial \tau_{12}^e}{\partial q} \right) \frac{\partial q}{\partial y}$$

$$\frac{\partial \tau_{12}}{\partial q} = \frac{\partial}{\partial q} (C_\mu f_v k^2 / \varepsilon) \Rightarrow \begin{cases} 2C_\mu f_v k / \varepsilon = 2\tau_{12} k^{-1} \\ -C_\mu f_v (k / \varepsilon)^2 = -\tau_{12} \varepsilon^{-1} \end{cases}$$

assuming $\text{Pr}^t \approx \text{Pr}$, hence $\text{RePr} \approx \text{RePr}^t = \text{Pe}$

$$\begin{aligned} [\text{JKK}]_e &= (\Delta x / 2, \text{Pe}^{-1}) (\) \{ \ } (-1) [\text{A211}] [\] \\ &+ (\Delta x / 2, \text{Pe}^{-1}, C_k^{-1}) (\) \{ \text{RET} \} (-1) [\text{A3011}] [\] \\ &+ (\Delta x, \text{Pe}^{-1}, C_k^{-1}) (\) \{ K \} (-1) [\text{A3110}] [\text{RET}, K^{-1}] \\ &+ (\Delta x) (\) \{ U \} (0) [\text{A3100}] [\text{TXY}, K^{-1}] \end{aligned}$$

PNS.12B GWS^h + θ TS TKE Closure Jacobian Coupling

Continuing with jacobians

$$\begin{aligned} [JKE]_e &= (-\Delta x / 2, \text{Pe}^{-1}, C_k^{-1}) () \{K\} (-1) [A3110] [\text{RET}, E^{-1}] \\ &\quad + (-\Delta x / 2) () \{U\} (0) [A3100] [TXY, E^{-1}] \\ &\quad + (\Delta x / 2) () \{ \} (1) [A200] [] \end{aligned}$$

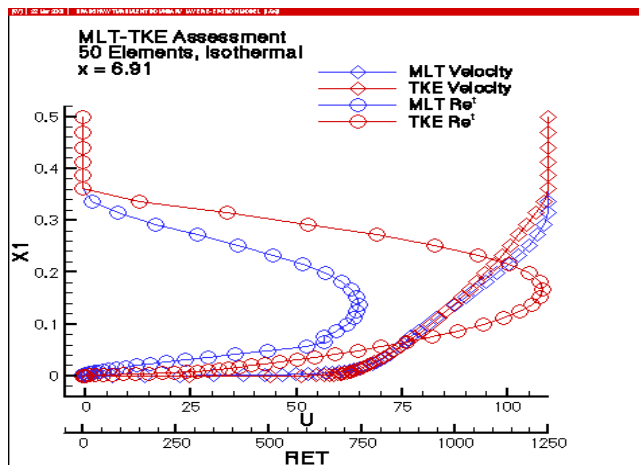
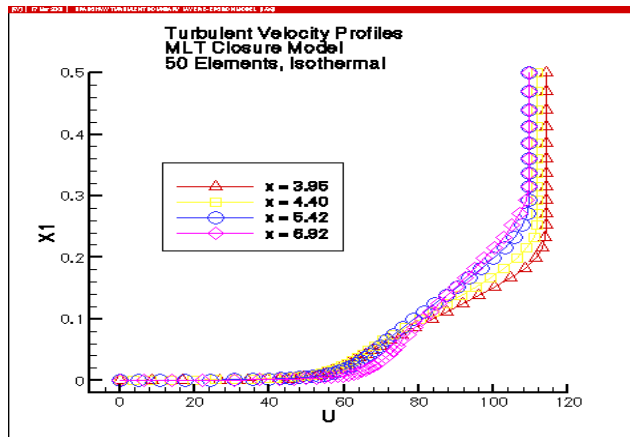
$$\begin{aligned} [JEK]_e &= (\Delta x, \text{Pe}^{-1}, C_\varepsilon^{-1}) () \{E\} (-1) [A3110] [\text{RET}, K^{-1}] \\ &\quad + (-\Delta x / 2, C_\varepsilon^1) (\text{FE1}) \{U\} (0) [A3100] [TXY, E / K^2] \\ &\quad + (\Delta x, C_\varepsilon^1) (\text{FE1}) \{U\} (0) [A3100] [TXY, (E / K)^2] \\ &\quad + (-\Delta x / 2, C_\varepsilon^2) (\text{FE2}) \{E\} (1) [A3000] [E / K^2] \end{aligned}$$

$$\begin{aligned} [JEE]_e &= (\Delta x / 2, \text{Pe}^{-1}) () \{ \} (-1) [A211] [] \\ &\quad + (\Delta x / 2, \text{Pe}^{-1}, C_\varepsilon^1) () \{\text{RET}\} (-1) [A3011] [] \\ &\quad + (\Delta x / 2, C_\varepsilon^2) (\text{FE2}) \{E / K\} (1) [A3000] [] \\ &\quad + (\Delta x / 2, C_\varepsilon^2) (\text{FE2}) \{E\} (1) [A3000] [K^{-1}] \end{aligned}$$

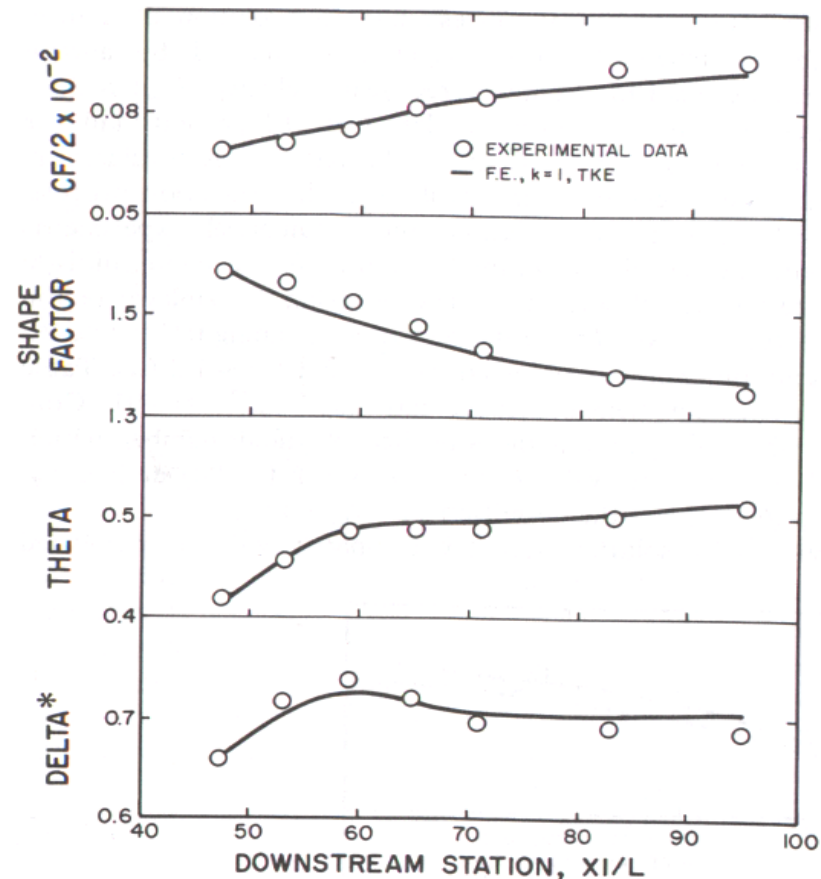
PNS.13 $GWS^h + \theta TS$ TKE BL, Validation

Validation, Bradshaw I 2400 experiment, $Re/L \approx 10^5$

MLT, TKE closure solutions



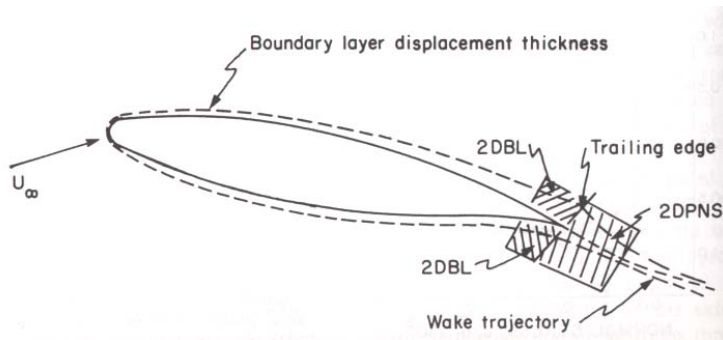
BL integral norm evolutions



PNS.14 Merging Turbulent Boundary Layers

Reynolds-ordered PNS PDE+BCs for merging BLs

Problem statement geometry



BL $\Rightarrow$ PNS theory modifications

DM BCs not valid for ODE on $\{V(y)\}$

$\Rightarrow \nabla \cdot \mathbf{u} = 0$ is now a differential constraint

DP_x remains as developed

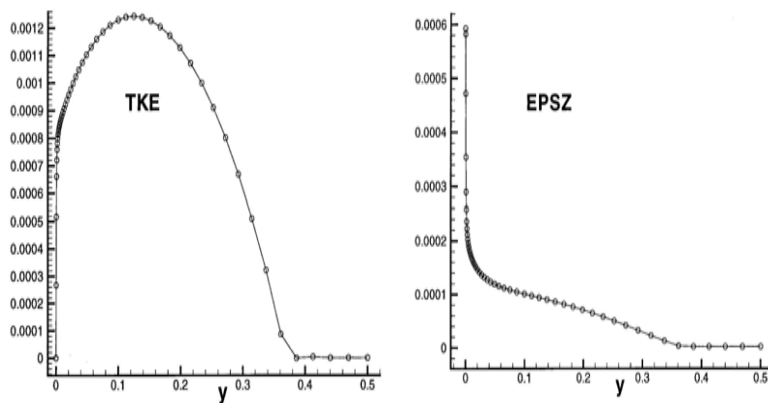
DP_y still $O(\delta)$, but must be included for BCs

DK, DE remain as developed

$\nabla \cdot DP$ yields pressure Poisson equation

$\Rightarrow$ complementary + particular solutions

BL distributions merging at TE



$\max(\partial k / \partial y, \partial \varepsilon / \partial y)|_{\partial \Omega} \Rightarrow$ interior to Ω !

requires Reynolds stress algebraic model

PNS.15 Merging Turbulent Boundary Layers, Reynolds Stress Model

Algebraic Reynolds stress model for uni-directional flows

$$\overline{u'_i u'_j} = k \alpha_{ij} - C_4 k^2 / \varepsilon S_{ij} - C_2 C_4 k^3 / \varepsilon^2 S_{ik} S_{kj}, \quad S_{ij} = \frac{\partial u_i}{\partial x_j} + \frac{\partial u_j}{\partial x_i}$$

for $n = 2$ BL:

$$\begin{aligned}\overline{u' u'} &= \overbrace{C_1 k - C_2 C_4 \frac{k^3}{\varepsilon^2} \left(\frac{\partial \bar{u}}{\partial y} \right)^2}^{O(\delta)} - \overbrace{2 C_4 \frac{k^2}{\varepsilon} \left(\frac{\partial \bar{u}}{\partial x} \right)}^{O(\delta^2)} \\ \overline{v' v'} &= C_3 k - C_2 C_4 \frac{k^3}{\varepsilon^2} \left(\frac{\partial \bar{u}}{\partial y} \right)^2 - 2 C_4 \frac{k^2}{\varepsilon} \left(\frac{\partial \bar{v}}{\partial y} \right) \\ \overline{w' w'} &= C_3 k \\ \overline{u' v'} &= C_2 \frac{k^2}{\varepsilon} \left(\frac{\partial \bar{u}}{\partial y} \right)\end{aligned}$$

closure model constants:

$$C_{01} \approx 2.8, \quad C_{02} \approx 0.45$$

$$C_1 \equiv \frac{22(C_{01} - 1) - 6(4C_{02} - 5)}{33(C_{01} - 2C_{02})}$$

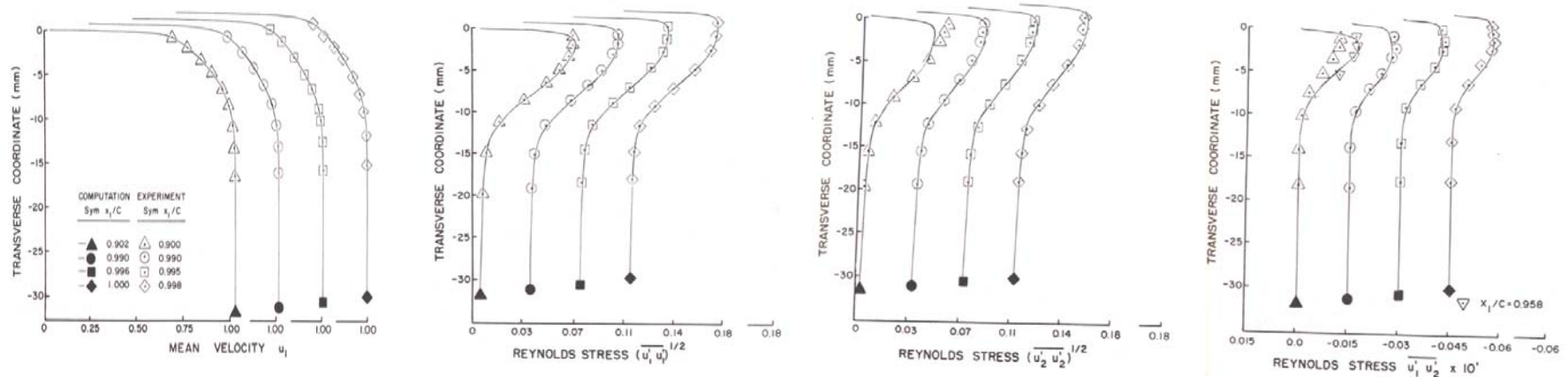
$$C_2 \equiv \frac{4(3C_{02} - 1)}{11(C_{01} - 2C_{02})}$$

$$C_3 \equiv \frac{22(C_{01} - 1) - 12(3C_{02} - 1)}{33(C_{01} - 2C_{02})}$$

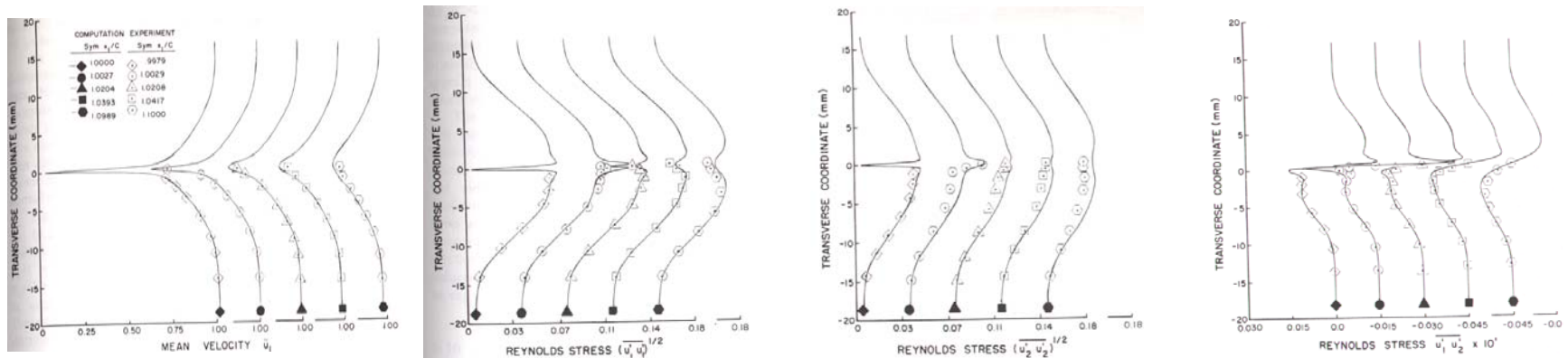
$$C_4 \equiv \frac{44C_{02} - 22C_{01}C_{02} - 128C_{02} - 36C_{02}^2 + 10}{165(C_{01} - 2C_{02})^2}$$

PNS.16 Validation, Merging Turbulent Boundary Layers

GWS^h+θTS BL comparison to data, $0.90 \leq x/c \leq 0.998$



GWS^h+θTS PNS wake comparison to data, $1.00 \leq x/c \leq 1.099$



PNS.17 Unidirectional 3-D Turbulent Flows

Algebraic Reynolds stress model for uni-directional flows

$$\overline{u_i' u_j'} = k \alpha_{ij} - C_4 k^2 / \varepsilon S_{ij} - C_2 C_4 k^3 / \varepsilon^2 S_{ik} S_{kj}, \quad S_{ij} = \frac{\partial u_i}{\partial x_j} + \frac{\partial u_j}{\partial x_i}$$

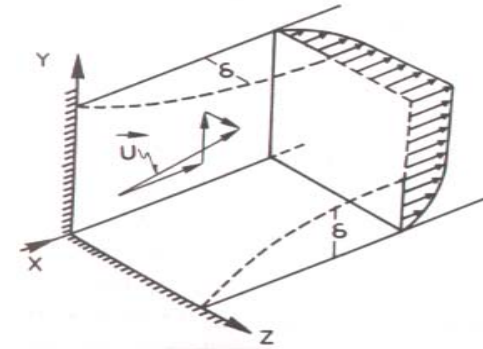
for $n = 2$ BL:

$$\begin{aligned} \overline{u' u'} &= \overbrace{C_1 k - C_2 C_4 \frac{k^3}{\varepsilon^2} \left(\frac{\partial \bar{u}}{\partial y} \right)^2}^{O(\delta)} - \overbrace{2 C_4 \frac{k^2}{\varepsilon} \left(\frac{\partial \bar{u}}{\partial x} \right)}^{O(\delta^2)} \\ \overline{v' v'} &= C_3 k - C_2 C_4 \frac{k^3}{\varepsilon^2} \left(\frac{\partial \bar{u}}{\partial y} \right)^2 - 2 C_4 \frac{k^2}{\varepsilon} \left(\frac{\partial \bar{v}}{\partial y} \right) \\ \overline{w' w'} &= C_3 k \\ \overline{u' v'} &= C_2 \frac{k^2}{\varepsilon} \left(\frac{\partial \bar{u}}{\partial y} \right) \end{aligned}$$

for $n = 3$ PNS:

$$\begin{aligned} \overline{u_1' u_1'} &= \overbrace{C_1 k - C_2 C_4 \frac{k^3}{\varepsilon^2} \left[\left(\frac{\partial \tilde{u}_1}{\partial x_2} \right)^2 + \left(\frac{\partial \tilde{u}_1}{\partial x_3} \right)^2 \right]}^{O(\delta)} - \overbrace{2 C_4 \frac{k^2}{\varepsilon} \left[\frac{\partial \tilde{u}_1}{\partial x_1} \right]}^{O(\delta^2)} \\ \overline{u_2' u_2'} &= C_3 k - C_2 C_4 \frac{k^3}{\varepsilon^2} \left[\frac{\partial \tilde{u}_1}{\partial x_2} \right]^2 - 2 C_4 \frac{k^2}{\varepsilon} \left[\frac{\partial \tilde{u}_2}{\partial x_2} \right] \\ \overline{u_3' u_3'} &= C_3 k - C_2 C_4 \frac{k^3}{\varepsilon^2} \left[\frac{\partial \tilde{u}_1}{\partial x_3} \right]^2 - 2 C_4 \frac{k^2}{\varepsilon} \left[\frac{\partial \tilde{u}_3}{\partial x_3} \right] \\ \overline{u_1' u_2'} &= -C_4 \frac{k^2}{\varepsilon} \left[\frac{\partial \tilde{u}_1}{\partial x_2} \right] - C_2 C_4 \frac{k^3}{\varepsilon^2} \left[\frac{\partial \tilde{u}_1}{\partial x_3} \left(\frac{\partial \tilde{u}_2}{\partial x_3} \right)^2 + \left(\frac{\partial \tilde{u}_3}{\partial x_2} \right)^2 + 2 \frac{\partial \tilde{u}_1}{\partial x_2} \left(\frac{\partial \tilde{u}_1}{\partial x_1} + \frac{\partial \tilde{u}_2}{\partial x_2} \right) \right] \\ \overline{u_1' u_3'} &= -C_4 \frac{k^2}{\varepsilon} \left[\frac{\partial \tilde{u}_1}{\partial x_3} \right] - C_2 C_4 \frac{k^3}{\varepsilon^2} \left[\frac{\partial \tilde{u}_1}{\partial x_2} \left(\frac{\partial \tilde{u}_2}{\partial x_3} \right)^2 + \left(\frac{\partial \tilde{u}_3}{\partial x_2} \right)^2 + 2 \frac{\partial \tilde{u}_1}{\partial x_3} \left(\frac{\partial \tilde{u}_1}{\partial x_1} + \frac{\partial \tilde{u}_3}{\partial x_3} \right) \right] \\ \overline{u_2' u_3'} &= -C_2 C_4 \frac{k^3}{\varepsilon^2} \left[\frac{\partial \tilde{u}_1}{\partial x_2} \frac{\partial \tilde{u}_1}{\partial x_3} \right] - C_4 \frac{k^2}{\varepsilon} \left(\frac{\partial \tilde{u}_2}{\partial x_3} + \frac{\partial \tilde{u}_3}{\partial x_2} \right) \end{aligned}$$

flow geometry



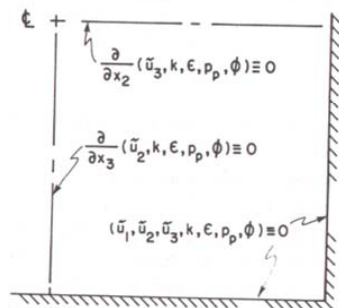
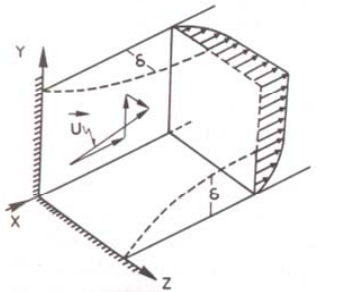
PNS.18 3D Turbulent Duct Flow, Validation

3D PNS algorithm based on PPNS formulation

$$DM : \nabla^h \cdot \bar{\rho} \tilde{\mathbf{u}}^h \approx 0 \Rightarrow \mathcal{L}(\phi) = -\nabla^2 \phi - \nabla \cdot \bar{\rho} \tilde{\mathbf{u}} = 0 + BCs$$

$$\nabla \cdot DP : \nabla \cdot \mathcal{L}(\tilde{\mathbf{u}}) = 0 \Rightarrow \mathcal{L}(p) = -\nabla \cdot \bar{\rho} \nabla p + s(\bar{\rho}, \tilde{\mathbf{u}}) = 0 + BCs$$

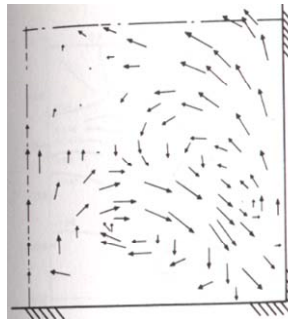
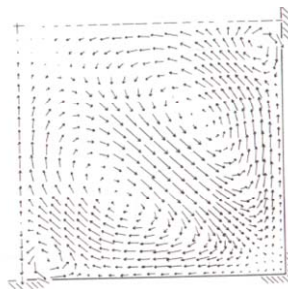
3D square duct flow, BCs



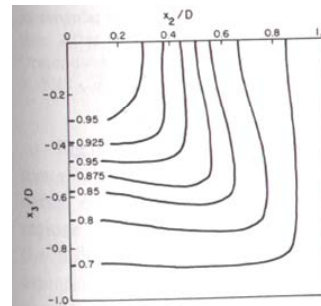
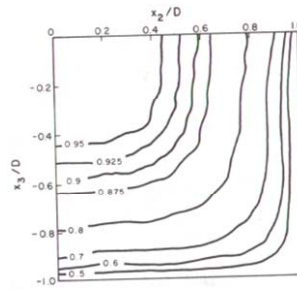
GWS^h + θ TS

experiment

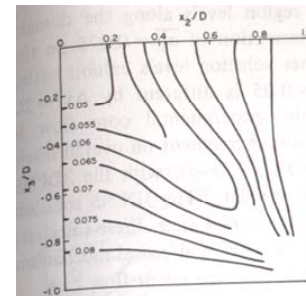
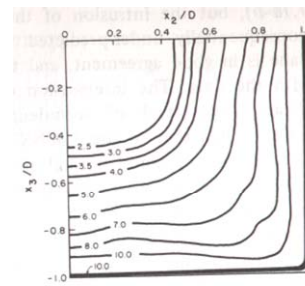
transverse $\bar{u}_k$



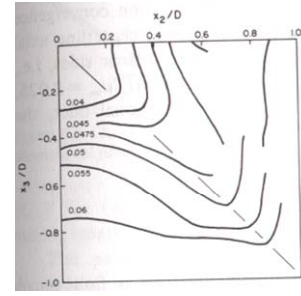
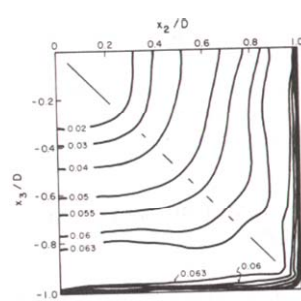
axial $\bar{u}_1$



stress $\overline{u_1' u_1'}$



stress $\overline{u_2' u_2'}$



PNS.19 Summary, Steady Turbulent Parabolic Navier-Stokes

Aerodynamic flows $\Leftrightarrow$ weak interaction

streamline shapes
flowfield is uni-directional
pressure impressed from farfield
large Reynolds number, $Re/L > 10^6$
viscous-turbulent effects strictly local
admits parabolizing steady NS equations



Validation exercises for MLT closure

Asymptotic convergence theory confirmed

$$\left| e^h(n\Delta x) \right|_E \leq Ch_e^{2\gamma} \left\| \text{data} \right\|_{L_2}^2 + C_x \Delta x^3 \left\| q_0 \right\|_{H^1}^2, \gamma = \min(k, r-1) \Rightarrow k$$

Validation exercises for TKE+low Re^t closure model

wall region meshing requirements are substantial
algebraic non-linear Reynolds stress model
transverse plane effects due to higher order phenomena